



<b>AEROSPACE RECOMMENDED PRACTICE</b>	<b>ARP1533™</b>	<b>REV. D</b>
	Issued            1996-01 Revised           2024-03	
	Superseding ARP1533C	
<b>(R) Procedure for the Analysis and Evaluation of Gaseous Emissions from Aircraft Engines</b>		

RATIONALE

SAE Aerospace Recommended Practice ARP1533D provides updated information regarding ambient engine inlet air composition and carbon balance values. The ambient air composition is updated to contemporary values. The carbon balance values are updated to be consistent with ARP1533B and ICAO Annex 16 Vol. II Appendix 3.

A new Appendix E is provided to establish a CO<sub>2</sub> EI calculation equation with ambient component correction.

Editorial and clarity updates have been implemented including updating equation format to SAE style.

TABLE OF CONTENTS

1.	SCOPE.....	3
1.1	ARP Sections .....	3
2.	REFERENCES.....	4
2.1	Applicable Documents .....	4
2.1.1	SAE Publications.....	4
2.1.2	Other References .....	4
2.2	Definitions .....	4
2.3	Nomenclature and Suggested Values.....	5
2.3.1	Subscripts .....	7
3.	INTRODUCTION.....	7
4.	COMBUSTION CHEMICAL EQUATION .....	9
4.1	Modified Combustion Chemical Equation .....	9
5.	MATRIX SOLUTION OF THE COMBUSTION CHEMICAL EQUATION.....	10
5.1	Basic Matrix.....	10
5.2	Measurements, Interferences, and Corrections.....	11
5.2.1	Interferences of Carbon Dioxide, Water and Oxygen .....	12
5.2.2	NO <sub>x</sub> Converter Efficiency.....	13
5.2.3	Sample Drying.....	13
6.	CALCULATION OF GASEOUS EMISSIONS PARAMETERS.....	15
7.	CALCULATION OF DATA QUALITY INDICATORS .....	17

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7.1	Sample Stability .....	17
7.2	Oxygen Balance .....	17
7.3	Carbon Balance .....	17
7.4	Fuel/Air Ratio Balance .....	18
7.5	NO/NO <sub>x</sub> Ratio .....	18
7.6	Trend Plots .....	18
8.	CALCULATION OF MEASUREMENT UNCERTAINTY .....	19
9.	NOTES .....	19
9.1	Revision Indicator .....	19
APPENDIX A	CALCULATION OF MEASUREMENT UNCERTAINTY .....	20
APPENDIX B	CALCULATION OF SAMPLE WATER CONTENT FROM FROST OR DEW POINT TEMPERATURE MEASUREMENTS .....	25
APPENDIX C	SAMPLE CALCULATIONS .....	28
APPENDIX D	DERIVATION OF EQUATIONS .....	36
APPENDIX E	CO <sub>2</sub> Emission Index Calculation .....	43
Table 1	Typical values of hydrocarbon fuel properties .....	8
Table 2	Properties of standard air (refer to Reference 2.1.2.8) .....	8
Table 3	Matrix A (see Equations 2 through 11 and 13) and Matrix B .....	11
Table 4	Expanded Matrix A and Matrix B .....	15

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## 1. SCOPE

SAE Aerospace Recommended Practice ARP1533 is a procedure for the analysis and evaluation of the measured composition of the exhaust gas from aircraft engines. Measurements of carbon monoxide, carbon dioxide, total hydrocarbon, and the oxides of nitrogen are used to deduce emission indices, fuel-air ratio, combustion efficiency, and exhaust gas thermodynamic properties. The emission indices (EI) are the parameters of critical interest to the engine developers and the atmospheric emissions regulatory agencies because they relate engine performance to environmental impact.

While this procedure is intended to guide the analysis and evaluation of the emissions from aircraft gas turbine engines (burning conventional hydrocarbon based liquid fuels), the methodology may be applied to the analysis of the exhaust products of any hydrocarbon/air combustor. Some successful applications include:

- Aircraft engine combustor development rig tests (aviation jet fueled)
- Stationary source combustor development rig tests (natural gas and diesel fueled)
- Afterburning military engine tests (aviation jet fueled)
- Internal combustion aircraft engine diagnostics (AVGAS fueled)

Each application may be characterized by very different measured emissions levels (parts per million versus percent by volume) but this common approach solves the same basic combustion chemical equation.

The matrix method of solving the combustion chemical equation is recommended because of all the potential variations in exhaust gas measurement requirements. Changes in the fuel type, addition of diluents, addition of measured species, and options for wet or dry basis measurements are most easily handled by revising individual matrix row equations. Matrix solution software is widely available on personal computers. However, derivation of the algebraic solution of the chemical equation is retained for traceability to previous versions of this document. This document also contains a section pertaining to data quality checks, measurement uncertainty, and water content calculations.

### 1.1 ARP Sections

This document is divided into the following sections:

2. References
  3. Introduction
  4. Combustion Chemical Equation
  5. Matrix Solution of the Combustion Chemical Equation
  6. Calculation of Gaseous Emissions Parameters
  7. Calculation of Data Quality Indicators
  8. Calculation of Measurement Uncertainty
- Appendix A - Calculation of Measurement Uncertainty
- Appendix B - Calculation of Sample Water Content from Frost or Dew Point Temperature Measurements
- Appendix C - Sample Calculations
- Appendix D - Derivation of Equations
- Appendix E - CO<sub>2</sub> Emissions Index Calculation

## 2. REFERENCES

### 2.1 Applicable Documents

The following publications form a part of this document to the extent specified herein. The latest issue of SAE publications shall apply. The applicable issue of other publications shall be the issue in effect on the date of the purchase order. In the event of conflict between the text of this document and references cited herein, the text of this document takes precedence. Nothing in this document, however, supersedes applicable laws and regulations unless a specific exemption has been obtained.

#### 2.1.1 SAE Publications

Available from SAE International, 400 Commonwealth Drive, Warrendale, PA 15096-0001, Tel: 877-606-7323 (inside USA and Canada) or +1 724-776-4970 (outside USA), [www.sae.org](http://www.sae.org).

ARP1256 Procedure for the Continuous Sampling and Measurement of Gaseous Emissions from Aircraft Turbine Engines

#### 2.1.2 Other References

- 2.1.2.1 Kilpatrick, D.A. (1981). *Calculation methodology for basic engine exhaust gaseous emission parameters* (Report No. National Gas Turbine Establishment Memorandum M81002).
- 2.1.2.2 Machta, L. and Hughes, E. (1970). Atmospheric oxygen in 1967 to 1970. *Science*, 168(3939), 1582–1584.
- 2.1.2.3 Heneghan, S.P. and Frayne, C.W. (2000). *Propagation of errors for combustion analysis using emission analyzer data* (Report No. AIAA-2000-0955).
- 2.1.2.4 Hardy, B. (1998, April). *ITS-90 formulations for vapor pressure, frost point temperature, dew point temperature, and enhancement factors in the range –100°C to +100°C* [Conference presentation]. The Proceedings of the Third International Symposium on Humidity & Moisture, Teddington, London, England.
- 2.1.2.5 BS 1339-1:2002. (2002). Humidity—Part 1: Terms, definitions and formulae.
- 2.1.2.6 JCGM 101:2008. (2008). Evaluation of measurement data—Supplement 1 to the “Guide to the expression of uncertainty in measurement”—Propagation of distributions using a Monte Carlo method.
- 2.1.2.7 Wagner, W. and Pruß, A. (2002). The IAPWS formulation 1995 for the thermodynamic properties of ordinary water substance for general and scientific use. *Journal of Physical and Chemical Reference Data*, 31(2).
- 2.1.2.8 Keeling, R.F. (2022). Atmospheric oxygen and carbon dioxide data for Alert, Cold Bay, Cape Kumukahi, La Jolla Pier, Mauna Loa Observatory, American Samoa, Cape Grim, Palmer Station and South Pole. <https://scripps2.ucsd.edu/data.html>.

### 2.2 Definitions

**AIRCRAFT GAS TURBINE ENGINE:** Any gas turbine engine used for aircraft propulsion or power generation, including those commonly called turbojet, turbofan, turboprop, or turboshaft type engines.

**COMBUSTION EFFICIENCY:** The percentage ratio of the energy actually released by the combustion process to the energy which would be realized if all the carbon in the fuel were oxidized to carbon dioxide and the hydrogen to water vapor.

**CONCENTRATION:** The volume fraction of the component of interest in the gas mixture expressed as volume percentage or as parts per million by volume.

**EMISSION INDEX:** The mass of emissions of a given constituent per unit mass of fuel, multiplied by 1000.

**FUEL-AIR RATIO:** The mass rate of fuel flow to the engine divided by the mass rate of dry airflow through the engine.

**GASEOUS EMISSIONS:** The gaseous products, formed during combustion, found in the exhaust from a gas turbine engine. In this document, the gaseous emissions measured and used to solve the combustion chemical equation are carbon monoxide, carbon dioxide, nitric oxide, nitrogen dioxide, and total hydrocarbon.

**MOLE FRACTION:** The volume concentration of a gas per unit volume of the gas mixture of which it is a part. In the context of the measurements discussed in this procedure, “volume concentration (or volume fraction)” and “molar concentration (or mole fraction)” are synonymous.

**NET HEAT OF COMBUSTION:** The energy released per unit mass of fuel due to its complete oxidation at constant pressure as measured by cooling the products to the initial temperature without condensation of the water vapor formed in the reaction.

**NO<sub>x</sub>:** Oxides of nitrogen, specifically, the sum of nitric oxide (NO) and nitrogen dioxide (NO<sub>2</sub>).

**PARTS PER MILLION VOLUME (ppmV):** The unit volume concentration of a gas per million unit volumes of the gas mixture of which it is a part. (Also applicable to mass measurements but only volume relationships are referred to in these procedures.)

**PARTS PER MILLION CARBON (ppmC):** The mole fraction of hydrocarbon in the exhaust multiplied by 10<sup>6</sup>. Thus, 1 ppm of methane is indicated as 1 ppmC. To convert ppm concentration of the hydrocarbon to an equivalent ppmC value, multiply ppm concentration by the number of carbon atoms per molecule of the gas. For example, 1 ppm propane translates as 3 ppmC hydrocarbon; 1 ppm hexane as 6 ppmC hydrocarbon.

**STANDARD AIR:** Simplified air composition used for EI calculation.

**TOTAL HYDROCARBON:** The total of hydrocarbon compounds of all classes and molecular mass.

### 2.3 Nomenclature and Suggested Values

[CH <sub>4</sub> ] <sub>b</sub>	Mole fraction of methane in dry inlet air
[CO]	Mole fraction concentration of carbon monoxide in the exhaust
[CO <sub>2</sub> ]	Mole fraction concentration of carbon dioxide in the exhaust
[CO <sub>2</sub> ] <sub>b</sub>	Mole fraction of carbon dioxide in dry inlet air (default value = 0.000417)
[C <sub>x</sub> H <sub>y</sub> ]	Mole fraction concentration of total hydrocarbon in the exhaust, expressed in ppmC
[H <sub>2</sub> O]	Mole fraction concentration of water in the exhaust
[N <sub>2</sub> ] <sub>b</sub>	Mole fraction of the balance of dry inlet air without carbon dioxide and oxygen. Mostly nitrogen with minor species. (default value = 0.790281)
[NO]	Mole fraction concentration nitric oxide in the exhaust
[NO <sub>2</sub> ]	Mole fraction concentration nitrogen dioxide in the exhaust
[NO <sub>x</sub> ]	Mole fraction concentration of the oxides of nitrogen in the exhaust
[O <sub>2</sub> ]	Mole fraction concentration of oxygen in the exhaust
[O <sub>2</sub> ] <sub>b</sub>	Mole fraction of O <sub>2</sub> in dry inlet air (default value = 0.209302)

[SO <sub>2</sub> ]	Mole fraction concentration of sulfur dioxide in the exhaust
[Z]	Mole fraction concentration of constituent Z in exhaust
C <sub>m</sub> H <sub>n</sub> O <sub>p</sub> N <sub>q</sub> S <sub>r</sub>	Chemical formula of 1 mole of hydrocarbon fuel
Diluent	A diluent such as nitrogen or water that is added to the combustion process
EI <sub>Z</sub>	Emission index of constituent Z, g/kg fuel (lb/1000 lb fuel)
f	Water vapor saturation pressure enhancement factor
F/A	Fuel-air ratio by mass
H <sub>c</sub>	Net heat of combustion of fuel at constant pressure J/kg (Btu/lb)
h	Water content of the inlet air, moles of water vapor per mole of dry inlet air
h <sub>sd</sub>	Water content of the semi-dry exhaust sample leaving the dryer, moles of water vapor per mole of dry sample gas
ICAO	International Civil Aviation Organization
J	Oxygen interference coefficient for effect of O <sub>2</sub> on the measurement of CO <sub>2</sub> (concentration factor)
K	Ratio of wet concentration to completely dry concentration
L	Interference coefficient for effect of CO <sub>2</sub> on the measurement of CO (zero shift)
L'	Interference coefficient for effect of CO <sub>2</sub> on the measurement of NO and NO <sub>x</sub> (concentration factor)
$\dot{m}_n$	Mass flow rate of fuel "n"
M	Interference coefficient for effect of H <sub>2</sub> O on the measurement of CO (zero shift)
M'	Interference coefficient for effect of H <sub>2</sub> O on the measurement of NO and NO <sub>x</sub> (concentration factor)
MAIR	Molecular mass of dry air (default value = 28.8544 g/mol)
Mc	Atomic mass of carbon = 12.0110
MH	Atomic mass of hydrogen = 1.0078
MN	Atomic mass of nitrogen = 14.0067
MO	Atomic mass of oxygen = 15.9994
MS	Atomic mass of sulfur = 32.0600
Mz	Atomic mass of constituent Z
P <sub>T</sub>	Total moles of exhaust products
PHYG	Gas sample pressure where the frost or dew point temperature measurement is made

PWV	Ideal water saturation vapor pressure at the measured frost or dew point temperature
PWVE	Effective water saturation vapor pressure at the gas sample pressure, PHYG
THYG	Frost or dew point temperature measured with a hygrometer
X	Moles of dry air/mole of fuel
$\alpha$	Atomic hydrogen-carbon ratio of the fuel = n/m
$\gamma, \phi$	Intermediate terms in the calculation of the water vapor pressure calculation enhancement factor, f
$\eta$	NO <sub>x</sub> converter efficiency
$\eta_b$	Combustion efficiency, %

### 2.3.1 Subscripts

d	Completely dry basis
m,n,p,q,r	Molar constants for fuel, C <sub>m</sub> H <sub>n</sub> O <sub>p</sub> N <sub>q</sub> S <sub>r</sub>
ms	Measured value
sd	Semidry basis
T	Combined (total) fuel properties, when fuels are mixed or diluted
w	Wet basis
x,y	Molar constants selected for the unburned hydrocarbon in exhaust, C <sub>x</sub> H <sub>y</sub>
z	Constituent Z

## 3. INTRODUCTION

The exhaust gas composition measurements required by this procedure include carbon monoxide (CO), carbon dioxide (CO<sub>2</sub>), nitric oxide (NO), oxides of nitrogen (NO<sub>x</sub>), total hydrocarbon (C<sub>x</sub>H<sub>y</sub>), as well as ambient air and sample dew point temperatures. The measured value of oxygen (O<sub>2</sub>) content is not required by the analysis but can be used for assessing data quality. Measurement of sulfur dioxide (SO<sub>2</sub>) is also not required but has been included herein to illustrate the addition of other species that may become significant and measurable. The instruments used for these measurements may be single-gas analyzers or single analyzers capable of measuring multiple gas compositions. Whichever the approach, each instrument must be fully characterized in terms of limitations in linearity, drift, interferences, repeatability, and biases. Some analyzers require a dried sample while others do not. Correct bookkeeping is required to ensure mixing wet basis, semi-dry and dry basis mole fractions does not occur. These issues are further explained in 5.2.

Ideally, the analysis and evaluation of emissions data should not require facility-measured parameters. However, when mixed fuels and diluents are used, facility metered flow rates are required to specify the number of moles of each reactant.

This procedure is valid for analysis of hydrocarbon/air exhaust products. Regardless of the hydrocarbon fuel; the chemical formula, molecular mass, and the lower heating value must be known. Typical values of these parameters for several fuels are provided in Table 1. However, precise values should be obtained from analysis of pre-and post-test fuel samples or from reference literature. The fuel chemical formula may be expanded to include the oxygen, nitrogen and sulfur in addition to carbon and hydrogen for a more rigorous balance of the reactants and products of combustion.

**Table 1 - Typical values of hydrocarbon fuel properties**

Fuel	Chemical Composition	H/C Ratio (n/m)	Lower Heating Value, Btu/lbm	Lower Heating Value, MJ/kg
Jet A	C <sub>11.6</sub> H <sub>22</sub>	1.897	18521	43.080
JP-5	C <sub>7.16</sub> H <sub>13.87</sub>	1.937	18300	42.567
JP-10	C <sub>10</sub> H <sub>16</sub>	1.6	18137	42.187
AVGAS	C <sub>7.68</sub> H <sub>16.8</sub>	2.187	18700	43.496
Natural Gas	C <sub>1.04</sub> H <sub>4.01</sub>	3.855	20680	48.102
#2 Diesel	C <sub>16</sub> H <sub>30</sub>	1.875	18318	42.608

Atmospheric oxygen and carbon dioxide levels in ambient air are, respectively, slowly decreasing and increasing over time. Appropriate standard air values for oxygen and carbon dioxide concentration are listed in Table 2 (these values will be updated in future document versions). The balance of standard air, [N<sub>2</sub>]<sub>b</sub> includes nitrogen and minor gas species. [N<sub>2</sub>]<sub>b</sub> is assumed to have equivalent molecular mass to N<sub>2</sub> and is calculated by:

$$[N_2]_b = 1 - [CO_2]_b - [O_2]_b$$

[CH<sub>4</sub>]<sub>b</sub> is considered negligible for the calculation of [N<sub>2</sub>]<sub>b</sub>

When large deviations from standard air are encountered (i.e., higher ambient carbon dioxide), actual concentrations should replace the dry air concentrations and [N<sub>2</sub>]<sub>b</sub> calculated.

A baseline sample of the inlet air should be analyzed, and the frost or dew point temperature measured.

**Table 2<sup>(1)</sup> - Properties of standard air (refer to Reference 2.1.2.8)**

Component	Chemical Formula	Molecular Mass	Percent by Volume
Air Balance [N <sub>2</sub> ] <sub>b</sub>	Assumed N <sub>2</sub>	28.0134	79.0281 (dry)
Oxygen [O <sub>2</sub> ] <sub>b</sub>	O <sub>2</sub>	31.9998	20.9302 (dry)
Carbon Dioxide [CO <sub>2</sub> ] <sub>b</sub>	CO <sub>2</sub>	44.0098	0.0417 (dry)
Water vapor	H <sub>2</sub> O	18.015	Must be measured
Total		28.8544 (dry)	100 (dry)

<sup>(1)</sup> Table 2 dry values are based on the 2019 yearly average of observational CO<sub>2</sub> and O<sub>2</sub> concentration data of stations around the world from the Scripps O<sub>2</sub> Program (refer to Reference 2.1.2.8) These data are from remote locations or other locations situated so that they represent averages over large portions of the globe rather than local background sources. O<sub>2</sub> is determined using 20.946% (refer to Reference 2.1.2.2) as the baseline.

Data quality checks have been incorporated into this procedure to encourage the use of near real time automated evaluation of the emissions measurement process. While there may be others, the checklist included in Section 7 has proven to aid in the early identification of emissions measurement problems.

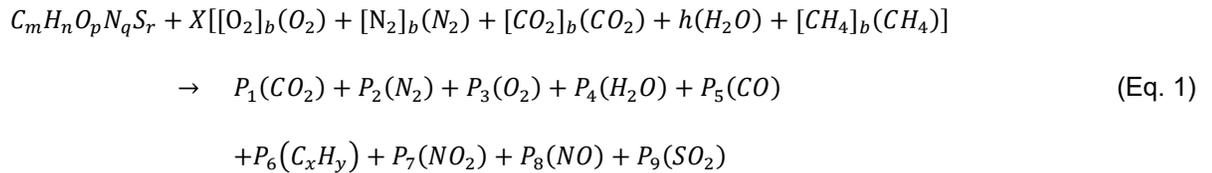
Calculations of emissions measurement uncertainties are increasingly necessary to show compliance with very low atmospheric emissions regulations. A methodology for the propagation of measurement errors in the calculations of the emissions indices (refer to Reference 2.1.2.3) is provided in Appendix A.

A method of calculating sample water content from measured frost and dew point temperatures (refer to Reference 2.1.2.4) is provided in Appendix B.

Two complete sample calculations are provided in Appendix C. The first calculation is the one presented in the version of ARP1533 issued in 1996. It is retained in this version for traceability of existing emissions analysis codes. The second calculation illustrates the expanded method of calculations with the matrix solution.

#### 4. COMBUSTION CHEMICAL EQUATION

The chemical equation for the combustion of 1 mole of hydrocarbon fuel and atmospheric air is:



where:

$P_1$  through  $P_9$  = number of moles of  $CO_2$ ,  $N_2$ ,  $O_2$ ,  $H_2O$ ,  $CO$ ,  $C_x H_y$ ,  $NO_2$ ,  $NO$ , and  $SO_2$  respectively

The reading of the hydrocarbon analyzer is expressed in ppmC, not in ppm, Therefore the value of  $P_6$  will depend on the assumption made regarding the composition of the unburned hydrocarbon  $C_x H_y$  in the exhaust gas. The balance of the combustion equation will also depend on the ratio between x and y.

This composition may be defined by at least three options between the SAE and ICAO methodologies:

- Equivalent fuel ( $CH_x$ ,  $x = 1$ ,  $y = n/m$ ) implies that the hydrocarbons in the exhaust gas have the same formulation as the fuel (old SAE method, used in the initial version of this document).
- Equivalent methane ( $CH_4$ ,  $x = 1$ ,  $y = 4$ ) implies that all the hydrocarbons in the exhaust gas are assumed to be methane in the absence of contrary evidence (ICAO method).
- Characterized fuel ( $C_x H_y$ ), used in the current version of this document, requires some knowledge of the composition of the hydrocarbons in the exhaust gas.

The analyst must clearly state which assumption was applied and must be careful about the consistency between the molecular mass and the concentration used to determine the emission index for Hc.

The combustion equation makes no allowance for the small amounts of hydrogen, free carbon, various species of hydrocarbon, and the oxides of nitrogen (other than nitric oxide and nitrogen dioxide) that may be present in the exhaust. Noble gases are unchanged during the combustion reaction. This procedure combines nitrogen, argon, and other trace species together for a total percent by volume of 79.0281. The water content of the inlet air, h, is expressed as moles of water vapor per mole of dry air.

The  $P_9(SO_2)$  term may be included in Equation 1 when the fuel has significant sulfur content. However,  $SO_2$  is not a required measurement for the calculation of emission indices.

##### 4.1 Modified Combustion Chemical Equation

Occasionally fuels are mixed or diluents are added to the fuel or gas stream during the development of new engines. When mixed fuels are used, where practical, it is recommended that the fuel mixture be analyzed to determine critical mixed fuel properties (e.g., H/C ratio of the fuel). The Section 4 approach may then be used directly. Where this approach is not sufficient, the user will need to determine what flows need to be monitored and what changes are required to the Section 4 analysis to accurately reflect their test configuration.

## 5. MATRIX SOLUTION OF THE COMBUSTION CHEMICAL EQUATION

### 5.1 Basic Matrix

Eleven simultaneous equations are required to solve for the eleven unknowns,  $P_1 - P_9$ ,  $P_T$  and  $X$ , in Equation 1. The procedure for writing the eleven simultaneous equations is the same whether multiple fuels and/or diluents are used or not. The equations for a single fuel and no diluents are defined as follows:

Carbon balance:

$$m + ([CO_2]_b + [CH_4]_b)X = P_1 + P_5 + xP_6 \quad (\text{Eq. 2})$$

Hydrogen balance:

$$n + (2h + 4[CH_4]_b)X = 2P_4 + yP_6 \quad (\text{Eq. 3})$$

Oxygen balance:

$$p + (2[O_2]_b + 2[CO_2]_b + h)X = 2P_1 + 2P_3 + P_4 + P_5 + 2P_7 + P_8 + 2P_9 \quad (\text{Eq. 4})$$

NOTE: The measured oxygen content is not used in the solution of the combustion equation.

Nitrogen balance:

$$q + 2[N_2]_bX = 2P_2 + P_7 + P_8 \quad (\text{Eq. 5})$$

Sulfur balance:

$$r = P_9 \quad (\text{Eq. 6})$$

Moles of carbon dioxide:

$$P_1 = P_T[CO_2]_w \quad (\text{Eq. 7})$$

Moles of carbon monoxide:

$$P_5 = P_T[CO]_w \quad (\text{Eq. 8})$$

Moles of total hydrocarbon:

$$xP_6 = P_T[C_xH_y]_w \quad (\text{Eq. 9})$$

where:

$[C_xH_y]$  = expressed in ppmC

Moles of the oxides of nitrogen:

$$P_7 + P_8 = P_T[NO_x]_w \quad (\text{Eq. 10})$$

Moles of nitric oxide:

$$P_8 = P_T[NO]_w \quad (\text{Eq. 11})$$

Moles of water:

$$P_4 = P_T[H_2O]_w \quad (\text{Eq. 12})$$

Where total number of moles of product:

$$P_T = P_1 + P_2 + P_3 + P_4 + P_5 + P_6 + P_7 + P_8 + P_9 \quad (\text{Eq. 13})$$

Equations 2 through 11 and 13 are solved for the constant terms then entered into the 11 x 11 Matrix A (Variables) and the 1 x 11 Matrix B (Constants) in Table 3. Row 1 from Equation 2 reads as  $P_1 + P_5 + xP_6 - ([CO_2]_b + [CH_4]_b)X = m$ . Row 10 from Equation 13 reads as  $-P_T + P_1 + P_2 + P_3 + P_4 + P_5 + P_6 + P_7 + P_8 + P_9 = 0$ . The inverse of Matrix A is multiplied by Matrix B to obtain the 1 x 11 Matrix C which contains the eleven unknowns,  $P_1 - P_9$ ,  $P_T$  and X. Equations 7 through 11 use concentrations on a wet basis. Measurements made on a semidry basis need to be converted to measurements on a wet basis.

The elements of Matrix C are wet basis.

**Table 3 - Matrix A (see Equations 2 through 11 and 13) and Matrix B**

Equation	Matrix A										Matrix B	
	$P_T$	$P_1$	$P_2$	$P_3$	$P_4$	$P_5$	$P_6$	$P_7$	$P_8$	$P_9$	X	Constants
2	0	1	0	0	0	1	x	0	0	0	$-[CO_2]_b - [CH_4]_b$	m
3	0	0	0	0	2	0	y	0	0	0	$-2h-4[CH_4]_b$	n
4	0	2	0	2	1	1	0	2	1	2	$-2[O_2]_b - 2[CO_2]_b - h$	p
5	0	0	2	0	0	0	0	1	1	0	$-2[N_2]_b$	q
6	0	0	0	0	0	0	0	0	0	1	0	r
7	$[CO_2]_w$	-1	0	0	0	0	0	0	0	0	0	0
8	$[CO]_w$	0	0	0	0	1	0	0	0	0	0	0
9	$[C_xH_y]_w$	0	0	0	0	0	-x	0	0	0	0	0
10	$[NO_x]_w$	0	0	0	0	0	0	-1	-1	0	0	0
11	$[NO]_w$	0	0	0	0	0	0	0	-1	0	0	0
13	-1	1	1	1	1	1	1	1	1	1	0	0

## 5.2 Measurements, Interferences, and Corrections

In a perfect world all gas analyzers would provide precise, unambiguous measurements of gas composition. In reality, depending on the type of analyzer being used, the emissions measurement must be concerned with:

- The interference of carbon dioxide and water on the measured concentrations of carbon monoxide and the oxides of nitrogen.
- The interference of oxygen on the measured concentration of carbon dioxide.
- The efficiency of the converter that converts the oxides of nitrogen to nitric oxide.
- The basis of measurement (semidry or wet).

This section of the procedure addresses these issues and modifies the equations in Matrix A. Effects such as that of oxygen upon carbon monoxide have been determined to be negligible for modern analyzers currently in use.

### 5.2.1 Interferences of Carbon Dioxide, Water and Oxygen

The measurement of a single gaseous species may be affected by the presence of other gaseous species. For example, carbon dioxide and water vapor in the exhaust interfere with the non-dispersive infrared measurement of CO and the chemiluminescent measurement of NO. There are two different types of interferences, depending on the effect they have on the measurement.

- With the “zero shift” interference effect, the interfering species creates an offset on the measurement, which does not vary with the concentration measured. This is the case for the interference of CO<sub>2</sub> and H<sub>2</sub>O on CO, for example. An interference coefficient is required to quantify the shift of the number of mole measured for species [A] due to the number of mole of interfering species [B] that is present.
- With the “concentration factor” interference effect (or sensitivity effect), the interfering species modifies the slope of the response of the analyzer: therefore, the effect is proportional to the concentration measured. This is the case for the interference of CO<sub>2</sub> and H<sub>2</sub>O on NO. An interference coefficient is required that quantifies the modification of the parts per volume measured for species [A] due to the parts per volume of interfering species [B] that is present.

The convention used in defining these effects is that the correction is added to the measured value (or added to unity and multiplied by the measured value where appropriate) to give the best estimate of the true value. Thus, an interference that reduces the reading is expressed as a positive interference effect. L, L', M, M', and J are the interference coefficients for analyzers that should be determined under test conditions. Example values of interference coefficients (refer to Reference 2.1.2.1) are provided below. These values are used in the sample calculations in Appendix C but are not representative of current analyzers performances. It is recommended to measure the real interference coefficients for each new instrument. If this is not possible, the uncertainties on the interference coefficients should be taken into account in the calculation of the uncertainties on the corrected concentrations (see Appendix A).

L = -1.3x 10<sup>-4</sup>, mole CO per mole CO<sub>2</sub> (zero shift effect)

M = -4.5 x 10<sup>-4</sup>, mole CO per mole H<sub>2</sub>O (zero shift effect)

L' = 0.14, percent of reading of NO per percent CO<sub>2</sub> (concentration factor effect)

M' = 0.28, percent of reading of NO per percent H<sub>2</sub>O (concentration factor effect)

J = 0.09, percent of reading of CO<sub>2</sub> per percent of O<sub>2</sub> (concentration factor effect)

Correct the CO measurement for the CO<sub>2</sub> and H<sub>2</sub>O interferences as follows:

$$P_5 = [CO]_{ms}P_T + LP_1 + MP_4 \quad (\text{Eq. 14})$$

Correct the NO<sub>x</sub> measurement for the CO<sub>2</sub> and H<sub>2</sub>O interferences as follows:

$$P_7 + P_8 = [NO_x]_{ms}P_T + L'[NO_x]_{ms}P_1 + M'[NO_x]_{ms}P_4 \quad (\text{Eq. 15})$$

Correct the NO measurement for the CO<sub>2</sub> and H<sub>2</sub>O interferences as follows:

$$P_8 = [NO]_{ms}P_T + L'[NO]_{ms}P_1 + M'[NO]_{ms}P_4 \quad (\text{Eq. 16})$$

Correct the CO<sub>2</sub> measurement for the O<sub>2</sub> interference as follows:

$$P_1 = [CO_2]_{ms}P_T + J[CO_2]_{ms}P_3 \quad (\text{Eq. 17})$$

To include these interference effects in the matrix solution:

For CO, Equation 14 becomes:

$$[CO]_{ms}P_T + LP_1 + MP_4 - P_5 = 0 \quad (\text{Eq. 18})$$

For NO<sub>x</sub>, Equation 15 becomes:

$$[NO_x]_{ms}P_T + L'[NO_x]_{ms}P_1 + M'[NO_x]_{ms}P_4 - P_7 - P_8 = 0 \quad (\text{Eq. 19})$$

For NO, Equation 16 becomes:

$$[NO]_{ms}P_T + L'[NO]_{ms}P_1 + M'[NO]_{ms}P_4 - P_8 = 0 \quad (\text{Eq. 20})$$

For CO<sub>2</sub>, Equation 17 becomes:

$$[CO_2]_{ms}P_T + J[CO_2]_{ms}P_3 - P_1 = 0 \quad (\text{Eq. 21})$$

### 5.2.2 NO<sub>x</sub> Converter Efficiency

A chemiluminescent analyzer can be used to measure both nitric oxide, NO, and the oxides of nitrogen, NO<sub>x</sub>. The nitrogen dioxide concentration, NO<sub>2</sub>, is assumed to be the difference between the measured values of NO<sub>x</sub> and NO. In the NO<sub>x</sub> setting, the entire sample flows through a converter, which converts the NO<sub>2</sub> to NO such that all of the NO<sub>x</sub> is measured as NO. In the NO setting, the converter is bypassed and only the sampled NO is measured. The NO<sub>x</sub> converter has an efficiency,  $\eta$ , typically between 0.9 and 1.0. Reference 2.1.1 gives the procedure for determining this efficiency. The measured value of NO<sub>x</sub> must be corrected for the quantity of NO<sub>2</sub> that was not converted to NO. From the definition of NO<sub>x</sub> converter efficiency:

$$[NO_x]_{ms} = \eta[NO_2] + [NO] \quad (\text{Eq. 22})$$

Moles of nitrogen dioxide:

$$P_7 = P_T[NO_2]_w \quad (\text{Eq. 23})$$

Moles of nitric oxide:

$$P_8 = P_T[NO]_w \quad (\text{Eq. 11})$$

Correct the NO<sub>x</sub> measurement for converter efficiency as follows:

$$[NO_x]_{ms} = \eta P_7 + P_8 \quad (\text{Eq. 24})$$

The NO<sub>x</sub> converter efficiency affects only P<sub>7</sub>. Combine Equations 19 and 24 to include the interference effects and the NO<sub>x</sub> converter efficiency in the matrix solution:

$$[NO_x]_{ms}P_T + L'[NO_x]_{ms}P_1 + M'[NO_x]_{ms}P_4 - \eta P_7 - P_8 = 0 \quad (\text{Eq. 25})$$

### 5.2.3 Sample Drying

Some gas analysis systems require a cold trap or membrane dryer in the sample line ahead of the analyzers. Since these dryers do not remove the water vapor completely, the concentrations after the dryer are referred to as semidry. The frost or dew point temperature at the dryer exit is measured and the water vapor saturation pressure is obtained from the appropriate tables. Appendix B provides a method of calculating sample water content based on Reference 2.1.2.4. This method applies to inlet air frost or dew point temperatures as well as frost or dew point temperatures measured before the gas analysis system dryers as long as the local sample pressure is used to correct the ideal vapor pressure.

The water content of the sample after the sample dryer is closely approximated for very dry samples as:

$$h_{sd} = [H_2O]_{sd} = \frac{P_{4sd}}{P_T - P_4 + P_{4sd}} \quad (\text{Eq. 26a})$$

From Equation 26a, we can also write:

$$\begin{aligned} P_{4sd} &= h_{sd} \times (P_T - P_4 + P_{4sd}) = h_{sd} \times (P_T - P_4) + h_{sd} \times P_{4sd} \\ P_{4sd} \times (1 - h_{sd}) &= h_{sd} \times (P_T - P_4) \\ P_{4sd} &= (P_T - P_4) \times \frac{h_{sd}}{1 - h_{sd}} \end{aligned} \quad (\text{Eq. 26b})$$

The number of moles of semidry sample is greater than the number of moles of dry sample by  $P_{4sd}$ :

$$P_T - P_4 + P_{4sd} = (P_T - P_4) \left(1 + \frac{h_{sd}}{1 - h_{sd}}\right) = (P_T - P_4) \times \frac{1}{1 - h_{sd}} \quad (\text{Eq. 27})$$

The semidry concentrations of CO, CO<sub>2</sub>, NO, NO<sub>x</sub>, H<sub>2</sub>O and O<sub>2</sub> are:

$$[CO]_{sd} = \frac{P_5}{P_T - P_4 + P_{4sd}} = \frac{P_5}{P_T - P_4} \times (1 - h_{sd}) \quad (\text{Eq. 28})$$

$$[CO_2]_{sd} = \frac{P_1}{P_T - P_4} \times (1 - h_{sd}) \quad (\text{Eq. 29})$$

$$[NO]_{sd} = \frac{P_8}{P_T - P_4} \times (1 - h_{sd}) \quad (\text{Eq. 30})$$

$$[NO_x]_{sd} = \frac{\eta P_7 + P_8}{P_T - P_4} \times (1 - h_{sd}) \quad (\text{Eq. 31})$$

$$[H_2O]_{sd} = \frac{P_{4sd}}{P_T - P_4} \times (1 - h_{sd}) = h_{sd} \quad (\text{Eq. 32})$$

$$[O_2]_{sd} = \frac{P_3}{P_T - P_4} \times (1 - h_{sd}) \quad (\text{Eq. 33})$$

Combining the interference and sample drying effects on the measured concentration of CO, Equation 8 in Matrix A is replaced by:

$$([CO]_{ms} + M h_{sd}) \times \frac{1}{1 - h_{sd}} \times (P_T - P_4) + L P_1 - P_5 = 0 \quad (\text{Eq. 34})$$

Combining the interference and sample drying effects on the measured concentration of CO<sub>2</sub>, Equation 7 in Matrix A is replaced by:

$$[CO_2]_{ms} \times \frac{1}{(1 - h_{sd})} \times (P_T - P_4) - P_1 + (J [CO_2]_{ms}) P_3 = 0 \quad (\text{Eq. 35})$$

Combining the interference and sample drying effects on the measured concentration of NO, Equation 11 in Matrix A is replaced by:

$$[NO]_{ms} \times (1 + (h_{sd} \times M')) \times \frac{1}{(1 - h_{sd})} \times (P_T - P_4) + L [NO]_{ms} P_1 - P_8 = 0 \quad (\text{Eq. 36})$$

Combining the interference, sample drying effects, and NO<sub>x</sub> converter efficiency on the measured concentration of NO<sub>x</sub>, Equation 10 in Matrix A is replaced by:

$$[NO_x]_{ms} \times (1 + (h_{sd} \times M')) \times \frac{1}{(1-h_{sd})} \times (P_T - P_4) + L'[NO_x]_{ms}P_1 - \eta P_7 - P_8 = 0 \quad (\text{Eq. 37})$$

The equations that account for the interferences and the sample drying process have been incorporated into Matrix A in Table 4 to provide a comprehensive solution of the eleven simultaneous equations.

**Table 4 - Expanded Matrix A and Matrix B**

Equation	Matrix A										Matrix B	
	P <sub>T</sub>	P <sub>1</sub>	P <sub>2</sub>	P <sub>3</sub>	P <sub>4</sub>	P <sub>5</sub>	P <sub>6</sub>	P <sub>7</sub>	P <sub>8</sub>	P <sub>9</sub>	X	Constants
2	0	1	0	0	0	1	x	0	0	0	-[CO <sub>2</sub> ] <sub>b</sub> -[CH <sub>4</sub> ] <sub>b</sub>	m
3	0	0	0	0	2	0	y	0	0	0	-2h-4- [CH <sub>4</sub> ] <sub>b</sub>	n
4	0	2	0	2	1	1	0	2	1	2	-2[O <sub>2</sub> ] <sub>b</sub> - 2[CO <sub>2</sub> ] <sub>b</sub> - h	p
5	0	0	2	0	0	0	0	1	1	0	-2[N <sub>2</sub> ] <sub>b</sub>	q
6	0	0	0	0	0	0	0	0	0	1	0	r
35	[CO <sub>2</sub> ] <sub>ms</sub> /(1-h <sub>sd</sub> )	-1	0	J[CO <sub>2</sub> ] <sub>ms</sub>	- [CO <sub>2</sub> ] <sub>ms</sub> /(1-h <sub>sd</sub> )	0	0	0	0	0	0	0
34	([CO] <sub>ms</sub> + h <sub>sd</sub> M) /(1-h <sub>sd</sub> )	L	0	0	-([CO] <sub>ms</sub> + h <sub>sd</sub> M) /(1-h <sub>sd</sub> )	-1	0	0	0	0	0	0
9	[C <sub>x</sub> H <sub>y</sub> ] <sub>ms</sub>	0	0	0	0	0	-x	0	0	0	0	0
37	(1+h <sub>sd</sub> ×M')×[NO <sub>x</sub> ] <sub>ms</sub> /(1-h <sub>sd</sub> )	L'[NO <sub>x</sub> ] <sub>ms</sub>	0	0	-(1+h <sub>sd</sub> .M')×[NO <sub>x</sub> ] <sub>ms</sub> /(1-h <sub>sd</sub> )	0	0	-η	-1	0	0	0
36	(1+h <sub>sd</sub> ×M')×[NO] <sub>ms</sub> /(1-h <sub>sd</sub> )	L'[NO] <sub>ms</sub>	0	0	-(1+h <sub>sd</sub> .M')×[NO] <sub>ms</sub> /(1-h <sub>sd</sub> )	0	0	0	-1	0	0	0
13	-1	1	1	1	1	1	1	1	1	1	0	0

## 6. CALCULATION OF GASEOUS EMISSIONS PARAMETERS

### 6.1 Conversion from Wet to Dry Basis

To convert product coefficients from the wet basis (w) to the completely dry basis (d), multiply the wet basis coefficient by K:

where:

$$K = \frac{P_T}{P_T - P_4} \quad (\text{Eq. 38})$$

$$[CO_2]_d = K \times \frac{P_1}{P_T} \quad (\text{Eq. 39})$$

$$[N_2]_d = K \times \frac{P_2}{P_T} \quad (\text{Eq. 40})$$

$$[O_2]_d = K \times \frac{P_3}{P_T} \quad (\text{Eq. 41})$$

$$[CO]_d = K \times \frac{P_5}{P_T} \quad (\text{Eq. 42})$$

$$[C_xH_y]_d = x \times K \times \frac{P_6}{P_T} \text{ (expressed in ppmC)} \quad (\text{Eq. 43})$$

$$[NO_2]_d = K \times \frac{P_7}{P_T} \quad (\text{Eq. 44})$$

$$[NO]_d = K \times \frac{P_8}{P_T} \quad (\text{Eq. 45})$$

6.2 The fuel-air ratio,  $F/A$ , is the ratio of the mass flow rate of the fuel ( $C_mH_n$ ) to dry air. Equation 46 ignores the presence of trace species in the fuel and the air:

$$F/A = \frac{(1 \text{ mole fuel})(\text{molecular wt. of fuel})}{(\text{moles dry air})(\text{molecular wt. of dry air})} = \frac{mM_C + nM_H}{X \times M_{AIR}} = \frac{m(M_C + \alpha M_H)}{X \times M_{AIR}} \quad (\text{Eq. 46})$$

6.3 The emissions index of constituent  $Z$ ,  $EI_Z$ , is the ratio of the mass of constituent  $Z$  to 1000 mass units of fuel consumed. It is commonly referred to as the mass of  $Z$  per 1000 mass units of fuel.

$$EI_Z = \left[ \frac{\text{mass rate of } Z}{\text{mass rate of fuel}} \right] \times 1000 = \left[ \frac{\text{moles of } Z}{\text{moles of fuel}} \right] \left[ \frac{\text{molecular wt. of } Z}{\text{molecular wt. of fuel}} \right] \times 1000 \quad (\text{Eq. 47})$$

Thus:

$$EI_{CO} = \frac{P_5 \times M_{CO} \times 10^3}{m(M_C + \alpha M_H)} \quad (\text{Eq. 48})$$

$$EI_{NO} = \frac{P_8 \times M_{NO_2} \times 10^3}{m(M_C + \alpha M_H)} \quad (\text{Eq. 49})$$

$$EI_{NO_x} = \frac{(P_7 + P_8) \times M_{NO_2} \times 10^3}{m(M_C + \alpha M_H)} \quad (\text{Eq. 50})$$

NOTE: The molecular mass used in the calculation of the emission index of nitric oxide and the oxides of nitrogen is the molecular mass of nitrogen dioxide.

$$EI_{C_xH_y} = \frac{P_6 \times M_{C_xH_y} \times 10^3}{m(M_C + \alpha M_H)} \quad (\text{Eq. 51})$$

where:

$$M_{C_xH_y} = x \times M_C + y \times M_H$$

NOTE: The hydrocarbon emission index calculated in ICAO Annex 16 Volume II (in Appendix 3) is based on the assumption that the exhaust hydrocarbon can be expressed as  $CH_4$ . The reader should be aware that if the assumed composition of the exhaust hydrocarbon is such that  $y/x \neq 4$ , then Equation 51 will yield an emission index different from the one obtained with the ICAO method.

6.4 The combustion efficiency,  $\eta_b$ , is calculated on an enthalpy basis by subtracting the inefficiencies due to unburned hydrocarbon and CO from 100%. It neglects the effects of  $NO_x$  and  $H_2$  and the dissociation of combustion products.

In SI units:

$$\eta_b = \left[ 1.00 - 10109 \left( \frac{EI_{CO}}{H_c} - \frac{EI_{C_xH_y}}{1000} \right) \right] \times 100 \quad (\text{Eq. 52})$$

where:

$H_c$  = net heat of combustion of fuel (sometimes referred to as the lower heating value) in J/kg

In Imperial units:

$$\eta_b = \left[ 1.00 - 4.346 \left( \frac{EI_{CO}}{H_c} - \frac{EI_{CxHy}}{1000} \right) \right] \times 100 \quad (\text{Eq. 53})$$

where:

$H_c$  = net heat of combustion of fuel (sometimes referred to as the lower heating value) in Btu/lb

Reference 2.1.2.1 is the source of much of the subject matter of this document. Refer to Reference 2.1.2.1 for more detailed discussions of the various details relating to the calculations.

## 7. CALCULATION OF DATA QUALITY INDICATORS

The expense associated with the measurement of emissions from aircraft engines can be minimized if instrumentation problems are quickly identified and corrected. Data quality checks have been incorporated into this procedure to encourage the use of near real time automated assessment of the emissions measurement process. While there may be others, those identified below have proven valuable in the early identification of emissions measurement problems.

### 7.1 Sample Stability

The standard deviation of a set of gaseous species measurements is an indicator that both the source and the analyzer were stable during the period of time when the emissions measurements were collected. For example, when  $n$  scans of gas analyzer data are collected over a 10 second interval, the standard deviation of the  $n$  samples about the average value should not exceed the instrument manufacturer's quoted repeatability. The target value for this data quality indicator should account for known measurement oscillations as well as instrumentation measurement precision.

$$[Z]_{SDEV} = \sqrt{\frac{\sum [Z]_{ms}^2 - \frac{(\sum [Z]_{ms})^2}{n}}{n-1}} \quad (\text{Eq. 54})$$

### 7.2 Oxygen Balance

Even though a measurement of the oxygen concentration is not required to solve the combustion chemical equation, if a measurement of semi-dry oxygen concentration (converted to wet or dry basis) is available, it may be compared to the calculated wet or dry basis oxygen concentration. The target value for the oxygen balance is 0.5%  $O_2$  or less. The oxygen analyzer is usually the last one to stabilize so an acceptable oxygen balance is another indication that the emission source and the  $CO_2$  and  $O_2$  analyzers were stable.

$$O_2 \text{ Balance} = [O_2]_{dry} - \frac{[O_2]_{measured \text{ semidry}}}{1-h_{sd}} \quad (\text{Eq. 55})$$

### 7.3 Carbon Balance

Solution of the chemical equation results in the P-terms which are the number of moles of the product species. The number of carbon atoms in the product species is forced to be equal to the number of carbon atoms in the reactant species. However, a comparison of the facility-metered carbon inlet flow to the carbon atoms specified by the P-terms is a method of validating the fuel chemical formula, the ability of the sampling system to capture a representative sample, the facility fuel and air flow meters and the  $CO_2$  gas analyzer health. A target value for the carbon balance in a combustion rig facility is  $1.0 \pm 0.05$ . For an engine test with engine conditions above idle, the carbon balance should be  $1.0 \pm 0.1$  due to increased uncertainties in engine air flow/mixing and sampling. For idle engine conditions, the carbon balance should be  $1.0 \pm 0.15$  due to increased uncertainty of engine exhaust mixing.

Calculate the ratio of the number of carbon atoms flowing into the combustor to the number of carbon atoms measured in the exhaust using Equation 56. Note that the ppm levels of carbon monoxide (P5) and unburned hydrocarbon (P6) have been ignored. Also note that  $W_{water}$  appears in the total flow rate term,  $(W_{fuel} + W_{air} + W_{water})$ . This accounts for any water injection. When the total flow into the combustor is known, the wet basis carbon dioxide mole fraction and the sample molecular mass should be used in the calculation of CARBAL:

$$CARBAL = \frac{\text{Calculated Carbon entering}}{\text{Measured Carbon leaving}} = \frac{\frac{W_{fuel} \times m + W_{air} \times [CO_2]_b}{MW_{fuel} + MW_{air}}}{\frac{(W_{fuel} + W_{air} + W_{water}) \times [CO_2]_w}{MW_{sample} \times 100}} \quad (\text{Eq. 56})$$

where:

$W_{fuel}$  = facility-metered fuel flow rate

$W_{air}$  = facility-metered air flow rate

$W_{water}$  = facility-metered water injection flow rate

$m$  = number of moles of carbon per mole of fuel from the most recent fuel analysis

$[CO_2]_b$  = number of moles of  $CO_2$  per mole of dry air

$MW_{sample}$  = molecular mass of the exhaust gas sample

$MW_{fuel}$  = molecular mass of the fuel

$MW_{air}$  = molecular mass of the inlet air

$[CO_2]_w$  = wet basis carbon dioxide mole fraction, percent by volume

#### 7.4 Fuel/Air Ratio Balance

A comparison of the fuel/air ratio calculation (see Equation 46) with the ratio of the facility-metered fuel and air mass flow rates is a higher-level indicator of emissions data quality, especially the  $CO_2$  measurement. Problems with the sample extraction process will be obvious here as well. A target value for the fuel/air ratio balance for a combustion rig facility is 5% or less. For an engine test with engine conditions above idle, the carbon balance should be  $1.0 \pm 0.1$  due to increased uncertainties in engine air flow/mixing and sampling. For idle engine conditions, the carbon balance should be  $1.0 \pm 0.15$  due to increased uncertainty of engine exhaust mixing.

$$FARBAL = \frac{(F/A_{emissions} - F/A_{facility})}{F/A_{facility}} \times 100 \quad (\text{Eq. 57})$$

#### 7.5 NO/NO<sub>x</sub> Ratio

For a given combustor, the ratio of NO/NO<sub>x</sub> will not vary over wide ranges of F/A. Once typical values are established (usually >0.7), changes in this ratio may be an indication of problems with either the NO or NO<sub>x</sub> analyzers.

$$NORatio = \frac{[NO]_w}{[NO_x]_w} \quad (\text{Eq. 58})$$

#### 7.6 Trend Plots

Trend plots that correlate calculated parameters are useful in detecting analyzer anomalies such as incorrect range settings or volume flow problems.

## 8. CALCULATION OF MEASUREMENT UNCERTAINTY

Calculations of emissions measurement uncertainties are necessary to show compliance with very low atmospheric emissions regulations. The propagation of errors analysis is a straightforward method of determining the measurement uncertainty of a calculated value,  $f(x_i)$  (e.g., the emission index), that is based on a set of input measured variables  $x_i$ . Of course, each of the input measurements is subject to its own error,  $\frac{\Delta x_i}{x_i}$ , depending upon the quality of the instrumentation used. Two methods of determining the uncertainty of the output calculations are presented here. The first is based on an analytic derivation, the details of which are summarized in Section A.1. This method yields a precise calculation that requires knowing the functional form, tedious differentiation, and assumes linearity over the range of errors for accuracy. The second method uses a stochastic process that estimates the uncertainty of the measurements. The stochastic method is simple to set up and yields correct results even for non-linear functional ranges. This method is discussed in Section A.2. Comparisons of propagated error using both methods, the analytic and stochastic, have shown their equivalency.

## 9. NOTES

### 9.1 Revision Indicator

A change bar (|) located in the left margin is for the convenience of the user in locating areas where technical revisions, not editorial changes, have been made to the previous issue of this document. An (R) symbol to the left of the document title indicates a complete revision of the document, including technical revisions. Change bars and (R) are not used in original publications nor in documents that contain editorial changes only.

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## APPENDIX A - CALCULATION OF MEASUREMENT UNCERTAINTY

## A.1 ANALYSIS OF THE MEASUREMENT UNCERTAINTY

To calculate the measurement uncertainties associated with the determination of emission indices, fuel/air ratio, and combustion efficiency from the matrix reductions provided in ARP1533, Equations A1 through A5 are analyzed utilizing the standard partial differential summation for error approach. The solutions via this approach are given in Equations A7 through A11.

For the purposes of that uncertainty analysis, one generally assumes that the uncertainty is associated only with the measurements of the mole fractions [CO], [CO<sub>2</sub>], [NO<sub>x</sub>], and [C<sub>x</sub>H<sub>y</sub>]. Other uncertainties exist but are usually small compared to the measured errors. If that is not the case, it is certainly also acceptable to use appropriate engineering judgement to estimate the total uncertainty in the values of [CO], [CO<sub>2</sub>], [NO<sub>x</sub>], and [C<sub>x</sub>H<sub>y</sub>].

Alternative techniques, such as adjoint approaches or Monte Carlo approaches, may also be used for a full uncertainty analysis of the matrix solution. Such approaches, however, are considered outside of the scope of this ARP.

$$EI_{CO} = \left( \frac{[CO]}{sum} \right) \left( \frac{10^3 \times M_{CO}}{M_C + \alpha M_H} \right) \left( 1 + \frac{X[CO_2]_b}{m} \right) \quad (\text{Eq. A1})$$

$$EI_{NOx} = \left( \frac{[NOx]}{sum} \right) \left( \frac{10^3 \times M_{NOx}}{M_C + \alpha M_H} \right) \left( 1 + \frac{X[CO_2]_b}{m} \right) \quad (\text{Eq. A2})$$

$$EI_{CxHy} = \left( \frac{[CxHy]}{sum} \right) \left( \frac{10^3 \times M_{CxHy}}{M_C + \alpha M_H} \right) \left( 1 + \frac{X[CO_2]_b}{m} \right) \quad (\text{Eq. A3})$$

$$\frac{F}{A} = \left( \frac{(1+h) \times (sum) - [CO_2]_b}{1 - \frac{\alpha}{4}(sum)} \right) \left( \frac{M_C + \alpha M_H}{M_{Air}} \right) \quad (\text{Eq. A4})$$

$$\eta_b = \left( 1.00 - 4.346 \times \frac{EI(CO)}{H_c} - \frac{EI(C_xH_y)}{1000} \right) \times 100 \quad (\text{Eq. A5})$$

where:

$$X = m \times \left( \frac{1 - \frac{\alpha}{4}(sum)}{(1+h) \times (sum) - [CO_2]_b} \right)$$

And “sum” is [CO] + [CO<sub>2</sub>] + [C<sub>x</sub>H<sub>y</sub>].

In general, for a dependent variable  $y = f(x_1, x_2, x_3, \dots, x_n)$ , the square of the expected relative uncertainty in  $y$  due to the relative uncertainties in the  $x_i$  ( $i = 1$  to  $n$ ) is given by Equation A6.

$$\left( \frac{\Delta y}{y} \right)^2 = \sum_i \left( \frac{\partial \ln y}{\partial \ln x_i} \right)^2 \left( \frac{\Delta x_i}{x_i} \right)^2 \quad (\text{Eq. A6})$$

where:

$$\frac{\Delta f}{f} = \text{relative uncertainty in } f$$

Applying Equation A6 to Equations A1 through A5, yields Equations A7 through A11, given in A.1.1.

## A.1.1 Solutions to Analytic Derivation of Uncertainty Measurements

CO uncertainty equation:

$$\left(\frac{\Delta EI_{CO}}{EI_{CO}}\right)^2 = \left(\frac{\partial \ln(EI_{CO})}{\partial \ln(CO)}\right)^2 \left(\frac{\Delta [CO]}{[CO]}\right)^2 + \left(\frac{\partial \ln(EI_{CO})}{\partial \ln(CO_2)}\right)^2 \left(\frac{\Delta [CO_2]}{[CO_2]}\right)^2 + \left(\frac{\partial \ln(EI_{CO})}{\partial \ln(C_xH_y)}\right)^2 \left(\frac{\Delta [C_xH_y]}{[C_xH_y]}\right)^2 \quad (\text{Eq. A7})$$

where:

$$\frac{\partial \ln(EI_{CO})}{\partial \ln(CO)} = \frac{[CO_2] + [C_xH_y]}{([CO] + [CO_2] + [C_xH_y])} + \frac{[CO_2]_b}{m + X[CO_2]_b} \times \frac{(-m) \times \left(1 + h - [CO_2]_b \frac{\alpha}{4}\right)}{[(1+h) \times ([CO] + [CO_2] + [C_xH_y]) - [CO_2]_b]^2} \times [CO]$$

$$\frac{\partial \ln(EI_{CO})}{\partial \ln(CO_2)} = \frac{-[CO_2]}{([CO] + [CO_2] + [C_xH_y])} + \frac{[CO_2]_b}{m + X[CO_2]_b} \times \frac{(-m) \times \left(1 + h - [CO_2]_b \frac{\alpha}{4}\right)}{[(1+h) \times ([CO] + [CO_2] + [C_xH_y]) - [CO_2]_b]^2} \times [CO_2]$$

$$\frac{\partial \ln(EI_{CO})}{\partial \ln(C_xH_y)} = \frac{-[C_xH_y]}{([CO] + [CO_2] + [C_xH_y])} + \frac{[CO_2]_b}{m + X[CO_2]_b} \times \frac{(-m) \times \left(1 + h - [CO_2]_b \frac{\alpha}{4}\right)}{[(1+h) \times ([CO] + [CO_2] + [C_xH_y]) - [CO_2]_b]^2} \times [C_xH_y]$$

NO<sub>x</sub> uncertainty equation:

$$\left(\frac{\Delta EI_{NO_x}}{EI_{NO_x}}\right)^2 = \left(\frac{\partial \ln(EI_{NO_x})}{\partial \ln(CO)}\right)^2 \left(\frac{\Delta [CO]}{[CO]}\right)^2 + \left(\frac{\partial \ln(EI_{NO_x})}{\partial \ln(CO_2)}\right)^2 \left(\frac{\Delta [CO_2]}{[CO_2]}\right)^2 + \left(\frac{\partial \ln(EI_{NO_x})}{\partial \ln(C_xH_y)}\right)^2 \left(\frac{\Delta [C_xH_y]}{[C_xH_y]}\right)^2 + \left(\frac{\partial \ln(EI_{NO_x})}{\partial \ln(NO_x)}\right)^2 \left(\frac{\Delta [NO_x]}{[NO_x]}\right)^2 \quad (\text{Eq. A8})$$

where:

$$\frac{\partial \ln(EI_{NO_x})}{\partial \ln(CO)} = \frac{-[CO]}{([CO] + [CO_2] + [C_xH_y])} + \frac{[CO_2]_b}{m + X[CO_2]_b} \times \frac{(-m) \times \left(1 + h - [CO_2]_b \frac{\alpha}{4}\right)}{[(1+h) \times ([CO] + [CO_2] + [C_xH_y]) - [CO_2]_b]^2} \times [CO]$$

$$\frac{\partial \ln(EI_{NO_x})}{\partial \ln(CO_2)} = \frac{-[CO_2]}{([CO] + [CO_2] + [C_xH_y])} + \frac{[CO_2]_b}{m + X[CO_2]_b} \times \frac{(-m) \times \left(1 + h - [CO_2]_b \frac{\alpha}{4}\right)}{[(1+h) \times ([CO] + [CO_2] + [C_xH_y]) - [CO_2]_b]^2} \times [CO_2]$$

$$\frac{\partial \ln(EI_{NO_x})}{\partial \ln(C_xH_y)} = \frac{-[C_xH_y]}{([CO] + [CO_2] + [C_xH_y])} + \frac{[CO_2]_b}{m + X[CO_2]_b} \times \frac{(-m) \times \left(1 + h - [CO_2]_b \frac{\alpha}{4}\right)}{[(1+h) \times ([CO] + [CO_2] + [C_xH_y]) - [CO_2]_b]^2} \times [C_xH_y]$$

$$\frac{\partial \ln(EI_{NO_x})}{\partial \ln(NO_x)} = 1$$

Hydrocarbon uncertainty equation:

$$\left(\frac{\Delta EI_{C_xH_y}}{EI_{C_xH_y}}\right)^2 = \left(\frac{\partial \ln(EI_{C_xH_y})}{\partial \ln(CO)}\right)^2 \left(\frac{\Delta [CO]}{[CO]}\right)^2 + \left(\frac{\partial \ln(EI_{C_xH_y})}{\partial \ln(CO_2)}\right)^2 \left(\frac{\Delta [CO_2]}{[CO_2]}\right)^2 + \left(\frac{\partial \ln(EI_{C_xH_y})}{\partial \ln(C_xH_y)}\right)^2 \left(\frac{\Delta [C_xH_y]}{[C_xH_y]}\right)^2 \quad (\text{Eq. A9})$$

where:

$$\frac{\partial \ln(EI_{C_xH_y})}{\partial \ln(CO)} = \frac{-[CO]}{([CO] + [CO_2] + [C_xH_y])} + \frac{[CO_2]_b}{m + X[CO_2]_b} \times \frac{(-m) \times (1 + h - [CO_2]_b \frac{\alpha}{4})}{[(1+h) \times ([CO] + [CO_2] + [C_xH_y]) - [CO_2]_b]^2} \times [CO]$$

$$\frac{\partial \ln(EI_{C_xH_y})}{\partial \ln(CO_2)} = \frac{-[CO_2]}{([CO] + [CO_2] + [C_xH_y])} + \frac{[CO_2]_b}{m + X[CO_2]_b} \times \frac{(-m) \times (1 + h - [CO_2]_b \frac{\alpha}{4})}{[(1+h) \times ([CO] + [CO_2] + [C_xH_y]) - [CO_2]_b]^2} \times [CO_2]$$

$$\frac{\partial \ln(EI_{C_xH_y})}{\partial \ln(C_xH_y)} = \frac{[CO] + [CO_2]}{([CO] + [CO_2] + [C_xH_y])} + \frac{[CO_2]_b}{m + X[CO_2]_b} \times \frac{(-m) \times (1 + h - [CO_2]_b \frac{\alpha}{4})}{[(1+h) \times ([CO] + [CO_2] + [C_xH_y]) - [CO_2]_b]^2} \times [C_xH_y]$$

Fuel Air Ratio uncertainty equation:

$$\left( \frac{\Delta(F/A)}{(F/A)} \right)_{Effluent} = \left( \frac{\partial \ln(F/A)}{\partial \ln(CO)} \right)^2 \left( \frac{\Delta CO}{CO} \right)^2 + \left( \frac{\partial \ln(F/A)}{\partial \ln(CO_2)} \right)^2 \left( \frac{\Delta CO_2}{CO_2} \right)^2 + \left( \frac{\partial \ln(F/A)}{\partial \ln(C_xH_y)} \right)^2 \left( \frac{\Delta C_xH_y}{C_xH_y} \right)^2 \quad (\text{Eq. A10})$$

where:

$$\frac{\partial \ln(F/A)}{\partial \ln(x_i)} = [x_i] \times \left[ \frac{4 \times (1+h) - \alpha \times [CO_2]_b}{[(1+h) \times ([CO] + [CO_2] + [C_xH_y]) - [CO_2]_b] \times [4 - \alpha \times ([CO] + [CO_2] + [C_xH_y])]} \right]$$

Combustion efficiency uncertainty equation:

$$\left( \frac{\Delta \eta_b}{\eta_b} \right)^2 = \left( \frac{\partial \ln(\eta_b)}{\partial \ln(EI_{CO})} \right)^2 \left( \frac{\Delta EI_{CO}}{EI_{CO}} \right)^2 + \left( \frac{\partial \ln(\eta_b)}{\partial \ln(EI_{C_xH_y})} \right)^2 \left( \frac{\Delta EI_{C_xH_y}}{EI_{C_xH_y}} \right)^2 \quad (\text{Eq. A11})$$

where:

$$\frac{\partial \ln(\eta_b)}{\partial \ln(EI_{CO})} = \frac{-EI_{CO}}{\left( \frac{100H_c}{4.346} - EI_{CO} - \frac{H_c}{4346} EI_{C_xH_y} \right)}$$

$$\frac{\partial \ln(\eta_b)}{\partial \ln(EI_{C_xH_y})} = \frac{-EI_{C_xH_y}}{\left( 1000 - \frac{4346}{H_c} EI_{CO} - EI_{C_xH_y} \right)}$$

Finally, the calculation of the uncertainty analysis for the input fuel/air ratio (that is the measurement of F/A based on facility mass flow data) is given by Equation A12. This is a second method of calculating the fuel air ratio. Comparing the input and combustion output F/A ratio is important because it shows an internal consistency in the experiment and establishes the crucial parameter describing the combustion condition.

The difference between the two calculations should be less than the square root of the sum of the uncertainties in Equations A4 and A12 (i.e., the normalized error should be lower than 1). If the calculations yield significantly different results, then the experimenter must decide which results are in error. It is our experience that the error is frequently associated with poor sampling of the combustion effluent. This is an excellent quality check on the entire system.

Input (facility) fuel air ratio uncertainty equation

$$\left( \frac{\Delta(F/A)}{(F/A)} \right)_{Input} = \left( \frac{\partial \ln(F/A)}{\partial \ln(F)} \right)^2 \left( \frac{\Delta F}{F} \right)^2 + \left( \frac{\partial \ln(F/A)}{\partial \ln(A)} \right)^2 \left( \frac{\Delta A}{A} \right)^2 \quad (\text{Eq. A12})$$

where:

$$\frac{\partial \ln(F/A)}{\partial \ln(F)} = 1$$

$$\frac{\partial \ln(F/A)}{\partial \ln(A)} = -1$$

This method yields a precise calculation of the uncertainty. However, to extend the analysis to the uncertainty of any other output variable, or to include the error associated with other input parameters (such as h) would require a detailed analysis of the fairly complicated partial differential equation. This formulation also assumes linear behavior of the function in the region of  $x_i$ . As a result, the formula does not give a correct answer when the function is sharply curved (generally unimportant in this analysis), or the errors are large. The stochastic process discussed in Section A.2 overcomes all these limitations.

## A.2 STOCHASTIC METHODOLOGY FOR DETERMINATION OF UNCERTAINTY

To overcome the difficulties associated with determining the propagation of uncertainties using analytical methods (applying Equation A6 to determine Equations A7 through A11), a numerical method can be used instead that performs this calculation in a statistical sense. A numerical method provides distributions of the output quantities, from which the required uncertainties can be determined. It takes as inputs at least the following:

- The measured variables [CO], [CO<sub>2</sub>], [C<sub>x</sub>H<sub>y</sub>], [NO], and [NO<sub>x</sub>]
- The scale used to make these mole fraction measurements
- The expected percent of full scale error associated with the measurement of variables [CO], [CO<sub>2</sub>], [C<sub>x</sub>H<sub>y</sub>], [NO], and [NO<sub>x</sub>]
- Measured value “h” determined from the hygrometer reading
- “[CO<sub>2</sub>]<sub>b</sub>”, the amount of CO<sub>2</sub> in dry air
- The formula for fuel C<sub>m</sub>H<sub>n</sub> and unburned hydrocarbon C<sub>x</sub>H<sub>y</sub>

The propagated uncertainty calculation is accomplished by adding “noise” to the input values and propagating it to the results using a Monte Carlo method (refer to Reference 2.1.2.6). To do this, each input variable is allowed to vary with a distribution about the input mean value (in most cases a Gaussian distribution). In practice, it is usually sufficient to add noise to only the mole fraction measurements, as in most real systems these are the major sources of error. The mean of the distribution is set equal to the input measured value, and the standard deviation of the distribution is set equal to the expected uncertainty of the variable (percentage of full scale uncertainty multiplied by the scale). There are many standard routines for generating Gaussian variables of a given mean and standard deviation. In Microsoft Excel®, such a Gaussian distributed random variable can be calculated by using the NORMINV function specifically, as shown in Equation A13.

$$\text{Variable} = \text{NORMINV}(\text{RAND}(), \mu, \sigma) \quad (\text{Eq. A13})$$

where:

RAND( ) assigns a random number (uniform distribution from 0 to 1)

The mean value,  $\mu$ , is set to the actual emissions analyzer reading

The standard deviation,  $\sigma$ , is set equal to the % full-scale uncertainty times the scale used

A set of N separate “noisy” random variables of the input parameters are generated, and N separate calculations of the output values are calculated. N must be large enough for the results to have a good repeatability: for example, 10,000 samples or more can be processed within a few minutes by current personal computers. That is the matrix equation discussed in 5.1 using the comprehensive matrix given in Table 4 must be solved N times, and the results used to calculate the appropriate emissions indices N times. The propagated uncertainty is then calculated as the standard deviation of the set of output parameters. N should be chosen so that random calculations provide sufficient precision to the uncertainty estimation, while not significantly slowing the calculation process.

Alternatively, Equations A1 through A5 can be used to calculate the errors by this methodology. Again, a set of N separate “noisy” mole fractions is generated, and the output parameter is calculated N times. The uncertainty is then calculated as the standard deviation of the N output calculations. The calculation time for a set of variables using the spreadsheet is under a second and would be significantly shorter in a “compiled” language. (Note that we have not done any careful calibration of the calculation speed, or optimization of the randomization calculations.)

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APPENDIX B - CALCULATION OF SAMPLE WATER CONTENT FROM FROST OR DEW POINT TEMPERATURE MEASUREMENTS

In this document, water content must be known in terms of the number of moles of water per mole of dry air ( $h$ ), the number of moles of water per mole of semidry sample ( $h_{sd}$ ), and the number of moles of water per mole of fuel ( $P_4$ ). These mole ratios are also referred to as molal ratios, volume ratios, or mixing ratio by volume (refer to Reference 2.1.2.5). Molal ratio is equal to the ratio of the vapor partial pressure to the pressure of dry air or sample. Therefore, the actual water vapor partial pressure must be known. Typically, the sample frost or dew point temperature (THYG) is measured with a hygrometer at the gas sampling system pressure (PHYG). Empirical correlations are used to calculate the ideal water vapor saturation pressure (PWV) from the measured frost or dew point temperature at standard atmosphere pressure (14.695 psia or 101325 Pa). An enhancement factor ( $f$ ) that accounts for varying gas sampling system pressure and the presence of other gas constituents is multiplied by the ideal saturation vapor pressure to obtain the effective water vapor saturation vapor pressure (PWVE). Increasing sample pressure increases the saturation temperature and decreasing sample pressure decreases the saturation temperature for constant water content.

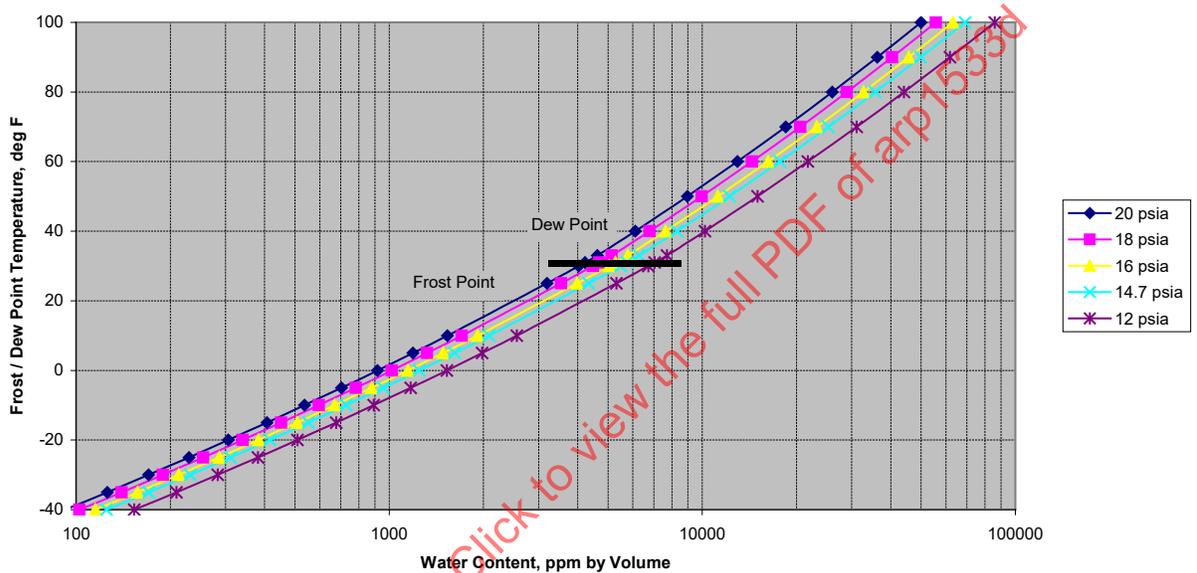


Figure B1

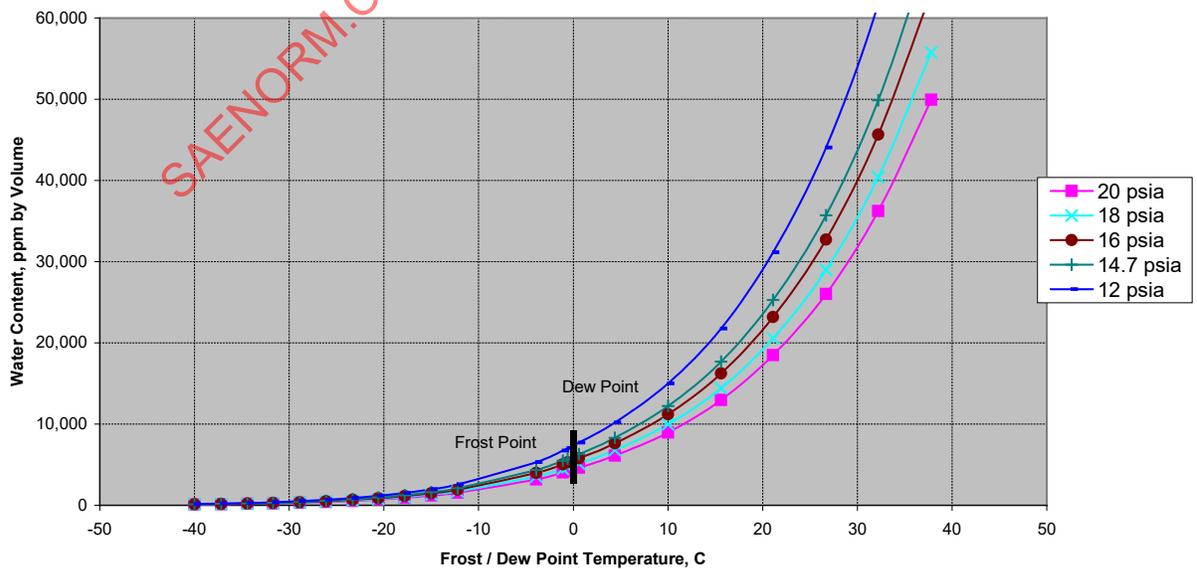


Figure B2

Many standards provide recommendations for dew point and water calculations. The empirical correlation presented by Hardy (refer to Reference 2.1.2.4), based on the ITS-90 temperature scale with traceability back to similar correlations of the IPTS-68 temperature scale that gained the largest international acceptance, is shown below as an example.

Measured dew point temperature, THYG, deg F converted to deg C and deg K

Ideal saturation vapor pressure at standard conditions, PWV, Pa

over water for THYG in deg K

$$PWV = \exp(\sum_{i=0}^6 (g_i \times THYG^{i-2}) + g_7 \times \ln(THYG)) \quad (\text{Eq. B1})$$

or over ice with THYG in deg K

$$PWV = \exp(\sum_{i=0}^4 (k_i \times THYG^{i-1}) + k_5 \times \ln(THYG)) \quad (\text{Eq. B2})$$

where:

g0	-2.8366E+03	
g1	-6.0281E+03	
g2	1.9543E+01	
g3	-2.7378E-02	
g4	1.6262E-05	
g5	7.0229E-10	
g6	-1.8680E-13	
g7	2.7150E+00	
k0		-5.8666E+03
k1		2.2329E+01
k2		1.3939E-02
k3		-3.4262E-05
k4		2.7041E-08
k5		6.7064E-01

Measured sample pressure, PHYG, psia converted to Pa

Enhancement factor, f:

$$\gamma = \sum_{i=0}^3 A_i \times THYG^i \quad \text{with THYG in deg C} \quad (\text{Eq. B3})$$

$$\phi = \exp(\sum_{i=0}^3 B_i \times THYG^i) \quad \text{with THYG in deg C} \quad (\text{Eq. B4})$$

	For Water -50 to 0 °C	For Water 0 to 100 °C	For Ice -100 to -50 °C	For Ice -50 to 0 °C
A <sub>0</sub>	3.62183x10 <sup>-4</sup>	3.53624x10 <sup>-4</sup>	9.8830022x10 <sup>-4</sup>	3.61345x10 <sup>-4</sup>
A <sub>1</sub>	2.6061244x10 <sup>-5</sup>	2.9328363x10 <sup>-5</sup>	5.7429701x10 <sup>-5</sup>	2.9471685x10 <sup>-5</sup>
A <sub>2</sub>	3.866777x10 <sup>-7</sup>	2.6168979x10 <sup>-7</sup>	8.9023096x10 <sup>-7</sup>	5.2191167x10 <sup>-7</sup>
A <sub>3</sub>	3.8268958x10 <sup>-9</sup>	8.5813609x10 <sup>-9</sup>	6.2038841x10 <sup>-9</sup>	5.0194210x10 <sup>-9</sup>
B <sub>0</sub>	-1.07604x10 <sup>1</sup>	-1.07588x10 <sup>1</sup>	-1.0415113x10 <sup>1</sup>	-1.07401x10 <sup>1</sup>
B <sub>1</sub>	6.3987441x10 <sup>-2</sup>	6.3268134x10 <sup>-2</sup>	9.1177156x10 <sup>-2</sup>	7.3698447x10 <sup>-2</sup>
B <sub>2</sub>	-2.6351566x10 <sup>-4</sup>	-2.5368934x10 <sup>-4</sup>	5.1128274x10 <sup>-5</sup>	-2.6890021x10 <sup>-4</sup>
B <sub>3</sub>	1.6725084x10 <sup>-6</sup>	6.3405286x10 <sup>-7</sup>	3.5499292x10 <sup>-6</sup>	1.5395086x10 <sup>-6</sup>

$$f = \exp \left[ \gamma \times \left( 1 - \frac{PWV}{PHYG} \right) + \phi \times \left( \frac{PHYG}{PWV} - 1 \right) \right] \quad (\text{Eq. B5})$$

Effective saturation vapor pressure, PWVE, Pa

$$PWVE = PWV \times f \quad (\text{Eq. B6})$$

Water content, ppm

$$h \text{ or } h_{sd} = \frac{PWVE}{(PHYG - PWVE)} \times 10^6 \quad (\text{Eq. B7})$$

An alternate empirical correlation is presented by Wagner and Pruss (refer to Reference 2.1.2.7), based on the 1995 IAPWS Formulation for the Thermodynamic properties of Ordinary Water Substance for General and Scientific Use. This formulation is the formulation adopted by the U.S. National Institute of Standards (NIST). The aforementioned approaches utilized to determine all relevant and appropriate water properties and water content relevant to gas sampling remain identical.

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## APPENDIX C - SAMPLE CALCULATIONS

Two sample calculations are provided. The first sample calculation uses data from earlier versions (1996-01 and before) of ARP1533 presented in the new format. The calculations use the ambient air composition values from earlier version ARP1533C. In addition to providing values to validate existing data reduction programs, this sample illustrates how to manipulate the matrix for:

- analysis of the fuel  $C_mH_nO_pN_qS_r$ , where  $p$ ,  $q$ , and  $r = 0$
- wet measurements of NO and  $NO_x$
- deleted measurements like  $SO_2$  where one row and one column of the matrix are not required

Note that perturbations in the input data and roundoff errors during the solution process may lead to differences in the results.

Sample Calculation #1

In this sample the fuel composition is estimated to be  $m = 9.5$ ,  $n = 19.0$ . The concentrations of  $C_xH_y$ , NO and  $NO_x$  are wet measurements, and the concentrations of CO and  $CO_2$  are semi-dry measurements. Matrix A has been simplified to account for these variations. This sample does not include data quality checks.

Average fuel formulation from chemical analysis of the pre- and post-test samples.

Fuel Formula	$C_{9.5} H_{19.0}$
Fuel lower heating value (FLHV), MJ/kg	43.566
Moles of carbon in fuel, m	9.5
Moles of hydrogen in fuel, n	19.0
Moles of oxygen in fuel, p	0
Moles of nitrogen in fuel, q	0
Moles of sulfur in fuel, r	0
Ratio of H to C in fuel	2.00

The expected composition of hydrocarbons in the exhaust gas is  $C_xH_y$  with  $x=1$  and  $y=2$  (SAE method).

Average air formulation from ambient air composition measurements.

Mole fraction of $O_2$ in air, $[O_2]_b$	0.20948	moles $O_2$ /mole dry air
Mole fraction of air balance, $[N_2]_b$	0.79020	moles $N_2$ /mole dry air
Mole fraction of $CO_2$ in air, $[CO_2]_b$	0.00032	moles $CO_2$ /mole dry air
Mole fraction of $CH_4$ in air, $[CH_4]_b$	0	moles $CH_4$ /mole dry air
Inlet air water content, h	0.00884	moles $H_2O$ /mole dry air

Instrument interferences from laboratory checks on instruments.

$CO_2$ interference on CO (L, mol CO / mol $CO_2$ ):	-1.3e-4
$H_2O$ interference on CO (M, mol CO / mol $H_2O$ ):	-4.5e-4
$CO_2$ interference on NO (L', %NO / % $CO_2$ ):	0.14
$H_2O$ interference on NO (M', %NO / % $H_2O$ ):	0.28
$O_2$ interference on $CO_2$ (J, % $CO_2$ / % $O_2$ ):	0.09
$NO_x$ converter efficiency (ETA):	0.95

NOTE: The oxygen interference coefficient,  $J = 0.09$ , assumes that the  $CO_2$  calibration gas uses  $N_2$  as the diluent.

Water content of inlet air and semi-dry sample.

Inlet air water content, h	0.00884	moles H <sub>2</sub> O/mole dry air
Semidry sample water content, h <sub>sd</sub>	0.00607	moles H <sub>2</sub> O/mole dry air

Measurement summary.

Species		Measurement
CO, ppm	semi-dry	500
CO <sub>2</sub> , percent	semi-dry	2
C <sub>x</sub> H <sub>y</sub> , ppmC	wet	225
NO, ppm	wet	9
NO <sub>x</sub> , ppm	wet	20

Simultaneous Equations: Matrix A

	P <sub>T</sub>	P <sub>1</sub>	P <sub>2</sub>	P <sub>3</sub>	P <sub>4</sub>	P <sub>5</sub>	P <sub>6</sub>	P <sub>7</sub>	P <sub>8</sub>	X
Eq. 2	0	1	0	0	0	1	1.0	0	0	-0.000320
Eq. 3	0	0	0	0	2	0	2.0	0	0	-0.01768
Eq. 4	0	2	0	2	1	1	0	2	1	-0.42844
Eq. 5	0	0	2	0	0	0	0	1	1	-1.5804
Eq. 35	0.020122	-1	0	0.0018	-0.020122	0	0	0	0	0
Eq. 34	0.000500	-1.30E-04	0	0	-5.00E-04	-1	0	0	0	0
Eq. 9	0.000225	0	0	0	0	0	-1.0	0	0	0
Eq. 25	0.000020	0.0000028	0	0	0.0000056	0	0	-0.95	-1	0
Eq. 20	0.000009	0.00000126	0	0	0.00000252	0	0	0	-1	0
Eq. 13	-1	1	1	1	1	1	1	1	1	0

Constants: Matrix B

	Constants
Eq. 2	9.5
Eq. 3	19.0
Eq. 4	0
Eq. 5	0
Eq. 35	0
Eq. 34	0
Eq. 9	0
Eq. 25	0
Eq. 20	0
Eq. 13	0

Results: Matrix C

		mole/mole fuel
Total (wet)	P <sub>T</sub>	469.01
carbon dioxide	P <sub>1</sub>	9.315
nitrogen	P <sub>2</sub>	363.51
oxygen	P <sub>3</sub>	82.382
water vapor	P <sub>4</sub>	13.463
carbon monoxide	P <sub>5</sub>	0.2267
total hydrocarbon	P <sub>6</sub>	0.1055
nitrogen dioxide	P <sub>7</sub>	0.00549
nitric oxide	P <sub>8</sub>	0.004267
air	X	460.03
Total (dry)	P <sub>T</sub> - P <sub>4</sub>	455.55

## Analysis Summary

Species	Formula	Measured Concentration	Wet Basis Concentration	Dry Basis Concentration	EI [Z] (kg Z/1000 Kg fuel)
oxygen, % calculated	O <sub>2</sub>	-	17.565	18.084	
carbon dioxide, % semi-dry	CO <sub>2</sub>	2.00	1.986	2.045	
carbon monoxide, ppm semi-dry	CO	500	483.4	497.6	47.65
nitrogen, % calculated	N <sub>2</sub>	-	77.51	79.796	
water vapor, % calculated	H <sub>2</sub> O	-	2.87	0	
total hydrocarbon, ppmC wet	CH <sub>2</sub>	225	225.0	231.6	11.11
nitrogen dioxide, ppm calculated	NO <sub>2</sub>	-	11.70	12.05	
nitric oxide, ppm wet	NO	9	9.10	9.37	1.47
oxides of nitrogen, ppm wet	NO <sub>x</sub>	20	20.80	21.42	3.37

## Gaseous Emissions Parameters

Emissions-derived fuel/air ratio	0.009998
Combustion efficiency, %	97.78

## Sample Calculation #2

In this sample the fuel composition is obtained from a comprehensive fuel analysis. Two cases are considered: in the first one, the concentration of C<sub>x</sub>H<sub>y</sub> is measured wet and the concentrations of CO, CO<sub>2</sub>, O<sub>2</sub>, NO, and NO<sub>x</sub> are semi-dry measurements. In the second one the concentrations of C<sub>x</sub>H<sub>y</sub>, NO, and NO<sub>x</sub> are wet measurements, and the concentrations of CO and CO<sub>2</sub> are semi-dry measurements. The matrices include a column and row to account for sulfur in the fuel and air even though SO<sub>2</sub> is not a measured product.

Average fuel formulation from chemical analysis of the pre- and posttest samples.

Fuel Formula:	<b>C7.158 H13.919 O0.00004 N0 S0.0012</b>
Fuel lower heating value, MJ/kg	43.148
Fuel Molecular Mass, g/mol	100.04
Moles of carbon in fuel, m	7.1576
Moles of hydrogen in fuel, n	13.9187
Moles of oxygen in fuel, p	0.00004
Moles of nitrogen in fuel, q	0
Moles of sulfur in fuel, r	0.0012
Ratio of H to C in fuel	1.945

The expected composition of hydrocarbons in the exhaust gas is C<sub>x</sub>H<sub>y</sub> with x=1 and y=1.945 (SAE method).

Average air formulation from periodic ambient air composition measurements.

Mole fraction of O <sub>2</sub> in air, [O <sub>2</sub> ] <sub>b</sub>	0.20687	moles O <sub>2</sub> /mole air
Mole fraction of air balance, [N <sub>2</sub> ] <sub>b</sub>	0.78036	moles N <sub>2</sub> /mole air
Mole fraction of CO <sub>2</sub> in air, [CO <sub>2</sub> ] <sub>b</sub>	0.00032	moles CO <sub>2</sub> /mole air
Mole fraction of CH <sub>4</sub> in air, [CH <sub>4</sub> ] <sub>b</sub>	0.0000037	moles CH <sub>4</sub> /mole air
Mole fraction of SO <sub>2</sub> in air,	0.000001	moles SO <sub>2</sub> /mole air
Ambient air dew point hygrometer operating pressure, PHYG <sub>1</sub>	97,900	Pa
Dew point temp of inlet air, THYG <sub>1</sub>	9.80	deg C
Effective vapor pressure, PWVE <sub>1</sub>	1220	Pa
Water content of inlet air, h <sub>1</sub>	0.01261	moles H <sub>2</sub> O/mole air

Instrument interferences from laboratory checks on instruments.

CO <sub>2</sub> interference on CO (L, mol CO / mol CO <sub>2</sub> ):	-1.3e-4
H <sub>2</sub> O interference on CO (M, mol CO / mol H <sub>2</sub> O):	-4.5e-4
CO <sub>2</sub> interference on NO (L', %NO / %CO <sub>2</sub> ):	0.14
H <sub>2</sub> O interference on NO (M', %NO / %H <sub>2</sub> O):	0.28
O <sub>2</sub> interference on CO <sub>2</sub> (J, %CO <sub>2</sub> / %O <sub>2</sub> ):	0.09
NO <sub>x</sub> converter efficiency (ETA):	0.975

Sample water content before and after the dryer.

Pre-dryer dew point hygrometer operating pressure, PHYG <sub>2</sub>	97,900	Pa
Dew point temp before dryer, THYG <sub>2</sub>	22.56	deg C
Effective vapor pressure, PWVE <sub>2</sub>	2,756	Pa
Pre-dryer water content, h <sub>2</sub>	0.02897	moles H <sub>2</sub> O/mole dry sample
Post-dryer dew point hygrometer operating pressure, PHYG <sub>3</sub>	97,900	Pa
Dew point temp after dryer, THYG <sub>3</sub>	-29.44	deg C
Effective vapor pressure, PWVE <sub>3</sub>	40.5	Pa
Post-dryer water content, h <sub>3</sub> or h <sub>sd</sub>	0.00041	moles H <sub>2</sub> O/mole dry sample

Facility-metered flow rates.

Engine power setting	79%
Fuel flow rate	0.110 kg/s
Air flow rate	13.45 kg/s

For the first case (NO/NO<sub>x</sub> measured in semi-dry basis), the results are as follow:

Measurement summary.

Species		Analyzer Range	Measurement	Standard Deviation
CO, ppm	semi-dry	1000	193.67	0.61
CO <sub>2</sub> , percent	semi-dry	10	1.77	0.12
O <sub>2</sub> , percent	semi-dry	25	18.61	0.21
C <sub>x</sub> H <sub>y</sub> , ppmC	wet	1000	85.5	1.30
NO, ppm	semi-dry	300	23.77	0.98
NO <sub>x</sub> , ppm	semi-dry	300	32.57	1.67

NOTE: A number of data scans have been recorded such that the reported measurement is the average value, and the standard deviation represents the stability of the emissions source and the analyzer during that time period.

## Simultaneous Equations: Matrix A

	P <sub>T</sub>	P <sub>1</sub>	P <sub>2</sub>	P <sub>3</sub>	P <sub>4</sub>	P <sub>5</sub>	P <sub>6</sub>	P <sub>7</sub>	P <sub>8</sub>	P <sub>9</sub>	X
Eq. 2	0	1	0	0	0	1	1.000	0	0	0	-0.00032
Eq. 3	0	0	0	0	2	0	1.945	0	0	0	-0.02523
Eq. 4	0	2	0	2	1	1	0	2	1	2	-0.42699
Eq. 5	0	0	2	0	0	0	0	1	1	0	-1.56072
Eq. 6	0	0	0	0	0	0	0	0	0	1	0
Eq. 35	0.017707	-1	0	0.001593	-0.017707	0	0	0	0	0	0
Eq. 34	0.000194	-0.00013	0	0	-0.000194	-1	0	0	0	0	0
Eq. 9	0.000086	0	0	0	0	0	-1.0	0	0	0	0
Eq. 37	0.000033	4.5598E-06	0	0	-0.000033	0	0	-0.975	-1	0	0
Eq. 36	0.000024	3.3278E-06	0	0	-0.000024	0	0	0	-1	0	0
Eq. 13	-1	1	1	1	1	1	1	1	1	1	0

## Constants: Matrix B

	Constants
carbon, Eq. 2	7.158
hydrogen, Eq. 3	13.919
oxygen, Eq. 4	0.0000
nitrogen, Eq. 5	0.0000
sulfur, Eq. 6	0.0012
carbon dioxide, Eq. 35	0
carbon monoxide, Eq. 34	0
total hydrocarbon, Eq. 9	0
oxides of nitrogen, Eq. 37	0
nitric oxide, Eq. 36	0
Total, Eq. 13	0

## Results: Matrix C

		mole/mole fuel
Total (wet)	P <sub>T</sub>	410.80
carbon dioxide	P <sub>1</sub>	7.1780
nitrogen	P <sub>2</sub>	317.76
oxygen	P <sub>3</sub>	73.681
water vapor	P <sub>4</sub>	12.0669
carbon monoxide	P <sub>5</sub>	0.0762
total hydrocarbon	P <sub>6</sub>	0.0351
nitrogen dioxide	P <sub>7</sub>	0.0036
nitric oxide	P <sub>8</sub>	0.0095
sulfur dioxide	P <sub>9</sub>	0.00120
air	X	407.20
Total (dry)	P <sub>T</sub> -P <sub>4</sub>	398.74

## Analysis Summary

Species	Formula	Measured Concentration	Measurement Standard Deviation	Wet Basis Concentration	Dry Basis Concentration	EI [Z] (kg/1000 kg fuel)
oxygen, % calculated	O <sub>2</sub>	-		17.94	18.48	
oxygen, % semi-dry	O <sub>2</sub>	18.61	± 0.21		19.15	
carbon dioxide, % semi-dry	CO <sub>2</sub>	1.77	± 0.12	1.75	1.80	
carbon monoxide, ppm semi-dry	CO	193.7	± 0.61	185.61	191.22	21.36
nitrogen, % calculated	N <sub>2</sub>	-		77.35	79.69	
water vapor, % calculated	H <sub>2</sub> O	-		2.94	0	
total hydrocarbon, ppmC wet	CH <sub>1.945</sub>	85.50	± 1.30	85.50	88.09	4.907
nitrogen dioxide, ppm calculated	NO <sub>2</sub>	-		8.79	9.05	1.660
oxides of nitrogen, ppm semi-dry	NO <sub>x</sub>	32.57	± 1.67	31.93	32.90	6.034
nitric oxide, ppm semidry	NO	23.77	± 0.98	23.14	23.84	4.374
sulfur dioxide, ppm calculated	SO <sub>2</sub>	-		2.92	3.01	0.769

## Gaseous Emissions Parameters

Emissions-derived fuel/air ratio	0.0086
Combustion efficiency	99.0089

NOTE: The emissions-derived fuel/air ratio calculated here is dry-basis and neglects the O<sub>p</sub>N<sub>q</sub>S<sub>r</sub> contribution to molecular mass.

## Data Quality Checks

## Sample stability from standard deviation

Species	Measured Concentration	Measurement Standard Deviation		Sample Stability Evaluation
oxygen, % semidry	18.61	±	0.21	1.128%, acceptable
carbon dioxide, % semi-dry	1.77	±	0.12	6.780%, marginally acceptable
carbon monoxide, ppm semidry	193.7	±	0.61	0.31%, acceptable
total hydrocarbon, ppmC wet	85.50	±	1.33	1.555%, acceptable
oxides of nitrogen, ppm semidry	32.57	±	1.67	5.12%, acceptable
nitric oxide, ppm semidry	23.77	±	0.98	4.12%, acceptable

## Balances and Ratios

Quality check	Value	Target and Evaluation
Oxygen Balance	-0.14	Within $\pm 0.5\%$ , acceptable
Carbon Balance	0.97	Within $\pm 0.1$ of 1.000, acceptable
F/A Balance	5.2	Outside $\pm 10/15\%$ for engines; $\pm 5\%$ for combustion rigs, check for problems
NO/NO <sub>x</sub> Ratio	0.72	Acceptable if it agrees well with trend for this engine.

For the second case (NO/NO<sub>x</sub> measured in wet basis), the results are as follow:

## Measurement summary.

Species		Measurement
CO, ppm	semi-dry	193.67
CO <sub>2</sub> , percent	semi-dry	1.77
O <sub>2</sub> , percent	semi-dry	18.61
C <sub>x</sub> H <sub>y</sub> , ppmC	wet	85.5
NO, ppm	wet	23.77
NO <sub>x</sub> , ppm	wet	32.57

## Simultaneous Equations: Matrix A

	P <sub>T</sub>	P <sub>1</sub>	P <sub>2</sub>	P <sub>3</sub>	P <sub>4</sub>	P <sub>5</sub>	P <sub>6</sub>	P <sub>7</sub>	P <sub>8</sub>	P <sub>9</sub>	X
Eq. 2	0	1	0	0	0	1	1	0	0	0	-0.00032
Eq. 3	0	0	0	0	2	0	1.945	0	0	2	-0.02523
Eq. 4	0	2	0	2	1	1	0	2	1	0	-0.42699
Eq. 5	0	0	2	0	0	0	0	1	1	0	-1.56072
Eq. 6	0	0	0	0	0	0	0	0	0	1	0
Eq. 35	0.017707	-1	0	0.001593	-0.0177073	0	0	0	0	0	0
Eq. 34	0.000194	-0.00013	0	0	-0.0001936	-1	0	0	0	0	0
Eq. 9	0.000086	0	0	0	0	0	-1	0	0	0	0
Eq. 25	0.000033	4.56E-06	0	0	9.1196E-06	0	0	-0.975	-1	0	0
Eq. 20	0.000024	3.33E-06	0	0	6.6556E-06	0	0	0	-1	0	0
Eq. 13	-1	1	1	1	1	1	1	1	1	1	0

Only four cells are modified, corresponding to the use of Equation 25 instead of Equation 37 (for NO<sub>x</sub>), and Equation 20 instead of Equation 36 (for NO).

## Constants: Matrix B (no changes)

	Constants
carbon, Eq. 2	7.158
hydrogen, Eq. 3	13.919
oxygen, Eq. 4	0.00004
nitrogen, Eq. 5	0.0000
sulfur, Eq. 6	0.0012
carbon dioxide, Eq. 35	0
carbon monoxide, Eq. 34	0
total hydrocarbon, Eq. 9	0
oxides of nitrogen, Eq. 25	0
nitric oxide, Eq. 20	0
Total, Eq. 13	0

Results: Matrix C

		mole/mole fuel
Total (wet)	P <sub>T</sub>	410,805
carbon dioxide	P <sub>1</sub>	7,1780
nitrogen	P <sub>2</sub>	317,76
oxygen	P <sub>3</sub>	73,680
water vapor	P <sub>4</sub>	12,062
carbon monoxide	P <sub>5</sub>	0,0762
total hydrocarbon	P <sub>6</sub>	0,0351
nitrogen dioxide	P <sub>7</sub>	0,0037
nitric oxide	P <sub>8</sub>	0,0099
sulfur dioxide	P <sub>9</sub>	0,0012
air	X	407,204
Total (dry)	P <sub>T</sub> -P <sub>4</sub>	398,743

Analysis Summary

Species	Formula	Measured Concentration	Wet Basis Concentration	Dry Basis Concentration	EI [Z] (kg/1000 kg fuel)
oxygen, % calculated	O <sub>2</sub>	-	17.94	18.48	
oxygen, % semi-dry	O <sub>2</sub>	18.61		19.15	
carbon dioxide, % semi-dry	CO <sub>2</sub>	1.77	1.75	1.80	
carbon monoxide, ppm semi-dry	CO	193.7	185.61	191.22	21.36
nitrogen, % calculated	N <sub>2</sub>	-	77.35	79.69	
water vapor, % calculated	H <sub>2</sub> O	-	2.94	0	
total hydrocarbon, ppmC wet	CH <sub>1.945</sub>	85.50	85.50	88.09	4.907
nitrogen dioxide, ppm calculated	NO <sub>2</sub>		9.12	9.40	1.724
oxides of nitrogen, ppm wet	NO <sub>x</sub>	32.57	33.15	34.15	6.264
nitric oxide, ppm wet	NO	23.77	24.02	24.75	4.540
sulfur dioxide, ppm calculated	SO <sub>2</sub>	-	2.92	3.01	0.769

Gaseous Emissions Parameters

Emissions-derived fuel/air ratio	0.0086
Combustion efficiency	99.01

NOTE: The emissions-derived fuel/air ratio calculated here is dry-basis and neglects the O<sub>p</sub>N<sub>q</sub>S<sub>r</sub> contribution to molecular mass.